

$$\Rightarrow \langle T \rangle + \frac{1}{2} \left\langle \sum_i \bar{F}_i \cdot \bar{x}_i \right\rangle = 0$$

$$\Rightarrow \langle T \rangle = -\frac{1}{2} \left\langle \sum_i \bar{F}_i \cdot \bar{x}_i \right\rangle$$

where $\langle T \rangle$ denote a long time average of T and similarly for $\langle \bar{F}_i \cdot \bar{x}_i \rangle$.

If $V(x) = ax^{n+1}$

\Rightarrow for a single particle

$$F(x) = -a(n+1)x^n$$

$$xF(x) = -(n+1)V(x)$$

$$\Rightarrow \langle T \rangle = \left(\frac{n+1}{2} \right) \langle V(x) \rangle$$

Conditions for closed orbits : \rightarrow

For a circular orbit we need

$$\dot{x} = 0$$

$$\Rightarrow E - V_e(x) \Big|_{x=x_0} = \text{an extremum}$$

$$\Rightarrow \frac{dV_e(x)}{dx} \Big|_{x=x_0} = 0$$

For stable periodic orbits we need

$$\frac{d^2V_e}{dx^2} \Big|_{x=x_0} > 0$$

$$V_e = V(r) + \frac{l^2}{2mr^2}$$

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$$\left. \frac{dV_e}{dr} \right|_{r=r_0} = \left. \frac{dV(r)}{dr} \right|_{r=r_0} - \frac{l^2}{mr_0^{2+1}} = 0$$

$$\Rightarrow \frac{l^2}{mr_0^{2+1}} = \left. \frac{dV}{dr} \right|_{r=r_0} = \frac{l^2}{mr_0^3}$$

$$\frac{d^2 V_e}{dr^2} = \left. \frac{d^2 V}{dr^2} \right|_{r_0} + \frac{3l^2}{mr_0^4} > 0$$

$$\Rightarrow \left. \frac{d^2 V}{dr^2} \right|_{r_0} + \frac{3}{r_0} \left. \frac{dV}{dr} \right|_{r_0} > 0$$

If $V(r) = \frac{+kr^{n+1}}{(n+1)}$

$$\Rightarrow \left[\frac{dV}{dr} \right]_{r_0} = \frac{+k(n+1)r_0^n}{n+1}$$

$$\left. \frac{d^2 V}{dr^2} \right|_{r_0} = \frac{+k(n+1)n r_0^{n-1}}{n+1}$$

\Rightarrow for stability we need

$$+(n+1)n + 3(n+1) > 0$$

$$+n + 3 > 0 \Rightarrow -(n+3) < 0$$

~~$$n > -3$$~~

Bertrand's Theorem: \rightarrow If $V(r) = \frac{kr^{n+1}}{n+1}$ then closed orbits are stable only for $n=1$ or $n=-2$.

Equation for the orbit r : \rightarrow

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Differential form

$$L = T - V(r)$$

$$= \frac{m}{2} (\dot{r}^2 + r^2 \dot{\theta}^2) - V(r)$$

$$\frac{d}{dt} \left(\frac{\partial L}{\partial \dot{r}} \right) - \frac{\partial L}{\partial r} = 0$$

$$\Rightarrow \frac{d}{dt} (m\dot{r}) + \frac{\partial V(r)}{\partial r} - m r \dot{\theta}^2 = 0$$

$$\dot{\theta} = l / (m r^2)$$

$$\Rightarrow m\ddot{r} + \frac{\partial V(r)}{\partial r} - \frac{l^2}{m r^3} = 0$$

$$\text{Now } \dot{\theta} = \frac{d\theta}{dt} = \frac{l}{m r^2}$$

$$\Rightarrow dt = \frac{m r^2}{l} d\theta$$

$$\therefore \frac{d}{dt} \left(\frac{d}{dt} (r) \right) = \frac{l^2}{m^2 r^2} \frac{d}{d\theta} \left(\frac{1}{r^2} \left(\frac{dr}{d\theta} \right) \right)$$

$$u = 1/r \Rightarrow r = 1/u, \quad \frac{d}{d\theta} \left(\frac{1}{u} \right) = -\frac{1}{u^2} \frac{du}{d\theta}$$

$$\therefore \ddot{r} = \frac{-l^2 u^2}{m^2} \frac{d^2 u}{d\theta^2}$$

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$$\therefore \frac{-l^2 u^2}{m} \frac{d^2 u}{d\theta^2} - \frac{l^2 u^3}{m} = +u^2 \frac{d(V(1/u))}{du}$$

$$\therefore \frac{d^2 u}{d\theta^2} + u = -\frac{m}{l^2} \frac{d[V(1/u)]}{du} \rightarrow (3.34)$$

Integral form: \rightarrow

We have $\dot{\theta} = l/(mr^2)$

$$\dot{r} = \left[\frac{2}{m} \left[E - V(r) - \frac{l^2}{2mr^2} \right] \right]^{1/2}$$

$$\therefore \frac{dr}{d\theta} = \dot{r}/\dot{\theta} = \frac{\dot{r} m r^2}{l}$$

$$\therefore \int_{\theta_0}^{\theta} d\theta = \theta - \theta_0 = \int_{r_0}^r \frac{l dr}{r^2 \left[(2m) \left[E - V(r) - \frac{l^2}{2mr^2} \right] \right]^{1/2}}$$

$$\therefore \theta = \theta_0 - \int_{u_0}^u \frac{du}{\sqrt{\frac{2mE}{l^2} - \frac{2mV}{l^2} - u^2}}$$

If we let $V(r) = ar^{n+1}$
 $V(1/u) = au^{-n-1}$

$$\therefore \theta = \theta_0 - \int_{u_0}^u \frac{du}{\sqrt{\frac{2m}{l^2}(E - au^{-n-1}) - u^2}}$$

↳ 3.39

$n = 1, -2, -3$ yield trigonometric functions.

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$n = 5, 3, 0, -4, -5, -7$ yield elliptic integrals.

The Kepler problem: \rightarrow

For this problem $n = -2$, $a = -k$

$$\therefore \theta = \theta_0 - \int_{u_0}^u \frac{du}{\sqrt{\frac{2mE}{l^2} + \frac{2mk}{l^2}u - u^2}}$$

which ~~evaluates~~ evaluates to

$$\theta = \theta_0 - \arccos \left[\frac{\left(\frac{l^2 u}{mk} - 1 \right)}{\sqrt{1 + \frac{2El^2}{mk^2}}} \right]$$

$$\therefore \frac{1}{r} = \frac{mk}{l^2} (1 + e \cos(\theta - \theta_0))$$

$$\text{where } e \equiv \sqrt{1 + \frac{2El^2}{mk^2}} \equiv e$$

Another simpler solution is to use ~~3.39~~ (3.34) with

$$V(r) = -k/r \Rightarrow V(1/u) = -ku$$

$$\Rightarrow \frac{d}{du} V(1/u) = -k$$

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$$\therefore \frac{d^2 u}{d\theta^2} + u = \frac{mk}{l^2}$$

$$\text{Let } z = u - \frac{mk}{l^2}$$

$$\Rightarrow \frac{d^2 z}{d\theta^2} + z = 0$$

$$\Rightarrow z = B \cos(\theta - \theta_0) \text{ where}$$

B and θ_0 are integration constants

$$\therefore u = \frac{mk}{l^2} (1 + \epsilon \cos(\theta - \theta_0))$$

where ϵ is to be determined

$$\frac{mk}{l^2} \epsilon \equiv B$$

$$\therefore \frac{1}{r} = \frac{mk}{l^2} (1 + \epsilon \cos(\theta - \theta_0))$$

The general equation for a conic section is

$$1/r = C [1 + \epsilon \cos(\theta - \theta_0)]$$

where $\epsilon = \text{eccentricity}$

$$e = \sqrt{1 + \frac{2El^2}{mk^2}}$$

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$e > 1 \iff E > 0 \iff$ hyperbola

$e = 1 \iff E = 0 \iff$ parabola

$e < 1 \iff E < 0 \iff$ ellipse

$e = 0 \iff E = -\frac{mk^2}{2l^2} \iff$ circle

Consider $e < 1$ case.

$$r = \frac{l^2}{mk(1 + e \cos(\theta - \theta_0))}$$

$r = r_{\min}$ when $\cos(\theta - \theta_0) = +1$

$r = r_{\max}$ when $\cos(\theta - \theta_0) = -1$

$$r_{\max} = \frac{l^2}{mk(1 - e)}, \quad r_{\min} = \frac{l^2}{mk(1 + e)}$$

The semimajor axis $a \equiv \frac{r_{\min} + r_{\max}}{2}$

$$= \frac{l^2}{mk} \left(\frac{1}{1 - e^2} \right) = \frac{l^2}{mk} \frac{1}{\frac{-2El^2}{mk^2}} = \frac{-k}{2E}$$

$$\therefore a = \frac{-k}{2E} \Rightarrow E = \frac{-k}{2a}$$

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$$\Rightarrow \epsilon = \sqrt{1 + \frac{(-k)l^2}{a/mk^2}} = \sqrt{1 - \frac{l^2}{mka}}$$

$$\therefore l^2/(mk) = a(1 - \epsilon^2)$$

$$\Rightarrow r = \frac{a(1 - \epsilon^2)}{1 + \epsilon \cos(\theta - \theta_0)}$$

For $\epsilon = 0$ we have ~~from~~ $r = \text{const.}$

$$\Rightarrow \dot{r} = 0, \quad \dot{\theta} = \frac{l}{mr^2} = \text{const.}$$

$$T = \text{const.}, \quad V = -k/r = \text{const.}$$

By Virial Theorem for $V = -k/r$

we have for a circular orbit

$$\langle T \rangle = T = + \frac{(-2+1)kV}{2} = \frac{-V}{2}$$

$$E = T + V = V/2$$

$$\therefore E = \frac{-k}{2r_0}$$

Motion in time: \rightarrow

What is $r = r(t)$, $\theta = \theta(t)$.

We have

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$$t = \int_{r_0}^r dr \left[\frac{2}{m} \left(E - V(r) - \frac{l^2}{2mr^2} \right) \right]^{-1/2}$$
$$= \left(\frac{m}{2} \right)^{1/2} \int_{r_0}^r \frac{dr}{\sqrt{E + \frac{k}{r} - \frac{l^2}{2mr^2}}}$$

We also have $2a = -\frac{k}{E}$

$$\text{and } \frac{l^2}{mk} = a(1 - \epsilon^2)$$

$$\therefore t = \left(\frac{m}{2k} \right)^{1/2} \int_{r_0}^r \frac{r dr}{\sqrt{r - \frac{r^2}{2a} - \frac{a(1 - \epsilon^2)}{2}}}$$

Choose r_0 as the perihelion distance and make the substitution

$$r = a(1 - \epsilon \cos \psi), \quad \psi = 0, \quad \forall r = r_0$$

$$\therefore t = \left(\frac{ma^3}{k} \right)^{1/2} [\psi - \epsilon \sin \psi]$$

Since $\psi = \psi(r)$ we have in principle solved for $r = r(t)$ through $t = t(\psi)$. ψ is called the eccentric anomaly. Similarly we have

$$r = a(1 - \epsilon \cos \psi) = \frac{a(1 - \epsilon^2)}{1 + \epsilon \cos(\theta - \theta_0)}$$

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[Note we chose $r_0 = a(1 - \epsilon)$
which is naturally satisfied
at $\theta = \theta_0$.]

$$\therefore \text{ we get } \cos(\theta - \theta_0) = \frac{\cos \psi - \epsilon}{1 - \epsilon \cos \psi}$$

which may be rewritten as

$$\tan\left(\frac{\theta - \theta_0}{2}\right) = \left(\frac{1 + \epsilon}{1 - \epsilon}\right)^{1/2} \tan\left(\frac{\psi}{2}\right),$$

So in principle we also have
 $t = t(\psi(\theta)) = t(\theta) \Rightarrow \theta = \theta(t)$.

Our solution does work $\forall \epsilon \neq 1$
As $\epsilon \rightarrow 1$, $a \rightarrow \infty$ for r to be finite
and non-zero.

So $\forall \epsilon = 1$ we have $E = 0$

$$\text{We use } \therefore m r^2 \dot{\theta} = l$$

$$\Rightarrow d\theta = \frac{l}{m r^2} dt$$

$$\therefore \int dt = \frac{m}{l} \int r^2 d\theta$$

$$\therefore t = \frac{l^3}{m k^2} \int_{\theta_0}^{\theta} d\theta [1 + \epsilon \cos(\theta - \theta_0)]^{-2}$$

This is true $\forall \epsilon \in [0, \infty]$

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Now consider $\epsilon = 1$

$$\text{Then } t = \frac{l^3}{mk^2} \int_{\theta_0}^{\theta} d\theta [1 + \cos(\theta - \theta_0)]^{-2}$$

$$1 + \cos(\theta - \theta_0) = 2 \cos^2\left(\frac{\theta - \theta_0}{2}\right)$$

$$\therefore t = \frac{l^3}{4mk^2} \int_{\theta_0}^{\theta} d\theta \left[\sec^4\left(\frac{\theta}{2}\right) \right]$$

$$= \frac{l^3}{4mk^2} \int_{\theta_0}^{\theta} d\theta \left[\sec^2\left(\frac{\theta - \theta_0}{2}\right) \left[1 + \tan^2\left(\frac{\theta - \theta_0}{2}\right) \right] \right]$$

$$= \frac{2l^3}{4mk^2} \int_0^{\tan\left(\frac{\theta - \theta_0}{2}\right)} (1 + x^2) dx$$

$$t = \frac{l^3}{2mk^2} \left[\frac{\tan(\theta - \theta_0)}{2} \right] \left[1 + \frac{1}{3} \tan^2\left(\frac{\theta - \theta_0}{2}\right) \right]$$

Here $-\pi < \theta < \pi$.

$$\text{As } t \rightarrow -\infty, \theta \rightarrow -\pi$$

$$t \rightarrow 0, \theta \rightarrow 0$$

$$t \rightarrow \infty, \theta \rightarrow \pi,$$

Now let's get back to the ellipse
 $0 < \epsilon < 1$.

$$t = \sqrt{\frac{ma^3}{k}} \int_0^{\psi} [1 - e \cos \psi] d\psi$$

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Choose $\psi = 2\pi \Rightarrow t = T = \text{period of the ellipse}$

$$\therefore T = 2\pi a^{3/2} (m/k)^{1/2}$$

$$\Rightarrow T^2 \propto a^3$$

\Rightarrow Kepler's third law. Note the caveat

$$m = \frac{m_1 m_2}{m_1 + m_2}$$

strictly speaking $T^2 \propto ma^3$

$$\therefore T^2 \propto \left(\frac{m_1 m_2}{m_1 + m_2} \right) a^3$$

$$m_2/m_1 \ll 1 \Rightarrow T^2 \propto m_1 a^3$$

$m_1 = \text{solar mass}$ is same \forall planets

Kepler's 2ND law $\frac{dA}{dt} = \text{constant}$

already proved earlier.

An conserved vector $\therefore \rightarrow$